

## Final Exam

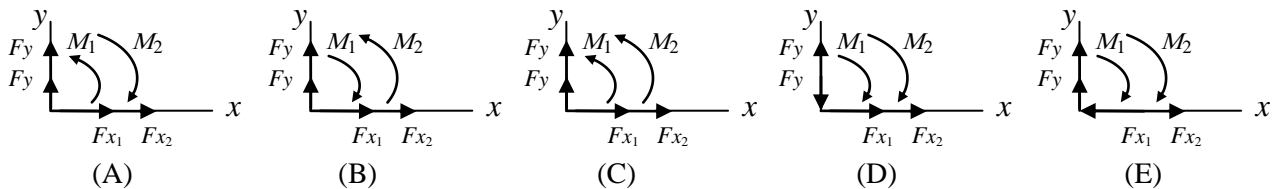
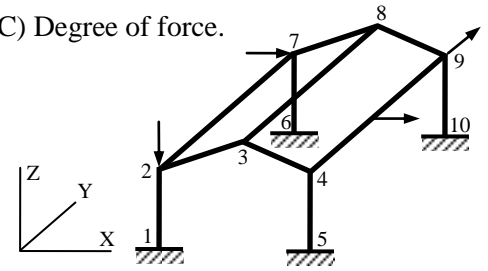
Total Marks: 60

No. of Questions: 3 (Attempt all questions)

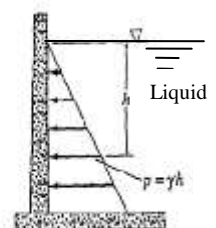
### Question (1): (20 Marks)

Choose the nearest correct answer (Put A, B, C, D or E in front of the statement number in your answer paper).

- The responsibility of the analytical model results lies on:
  - The company developed the software.
  - The computer used.
  - The input data.
  - The structural designer who used the software.
  - The structural designer who developed the software.
- The abbreviation "CAD" means:
  - Common Analysis Data.
  - Computer And Data.
  - Calculation And Design.
  - Computer-Aided Design.
  - Computer-Aided Data.
- The abbreviation "DOF" means:
  - Possible translations at nodes.
  - Possible rotations at nodes.
  - Degree of force.
  - Possible displacements at nodes.
  - Deformation of forces.
- The number of non-zero DOF for the shown space frame is:
  - 60
  - 30
  - 36
  - 18
  - 10
- The number of non-zero DOF per node 1 in the shown space frame is:
  - Zero
  - 1
  - 2
  - 3
  - 6
- The number of non-zero DOF per node 2 in the shown space frame is:
  - Zero
  - 3
  - 4
  - 6
  - 5
- If the axial deformation is neglected, the number of non-zero DOF per node 2 in the shown space frame is:
  - 4
  - 3
  - 5
  - 2
  - 6
- When there are loads between the nodes, the equilibrium equation of a plane frame is  $\{F\} = [K]\{\Delta\} + \{F^f\}$  where;
  - $\{F\}$  is the nodal forces.
  - $\{F^f\}$  is the nodal displacements.
  - $[K]$  is the element stiffness matrix.
  - $\{\Delta\}$  is the nodal forces.
  - $[K]$  is square of an order equal to the number of members.
- In 2D Analysis, ..... can be used.
  - only 2D elements
  - 1D, 2D and 3D elements
  - 2D and 3D elements
  - 1D and 2D elements
  - 1D and 3D elements
- Structures that cannot be modeled with the frame element are:
  - Space frames.
  - Space trusses.
  - Flat slabs.
  - Plane frames.
  - Beams.
- Stiffness is the property of an element which is defined as:
  - Displacement per unit area.
  - Displacement per unit force.
  - Force per unit displacement.
  - Force per unit displacement.
  - Force per unit mass.
- For plane frame element 1-2 (connecting joints 1 and 2), the positive sign of forces (forces and moments) is:



- One of the assumptions that the stiffness method is based on to analyze plane frames is:
  - Members will behave in non-linear and plastic manner.
  - Members (beams and columns) are straight with variable properties between joints.
  - Axial forces in members are very much less than the respective Euler buckling loads.
  - Applied loads may act out of the structure plane.
  - Deflections are sufficiently large for the changes of geometry to be ignored.
- In SAP, loads and properties of material are:
  - Fixed data.
  - Output data.
  - Input data.
  - Not important.
  - Results of the analysis
- In SAP, the ..... must be defined for (1D) plane frame elements.
  - Areas
  - Thicknesses
  - Sections
  - Volumes
  - Colors
- The triangle load applied on the shown vertical wall is called ..... load.
  - Earth pressure
  - Earthquake
  - Settlement
  - Temperature
  - Hydrostatic



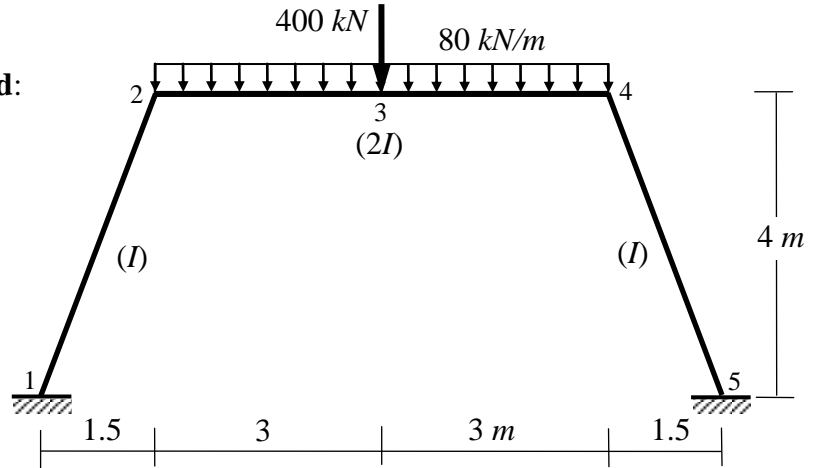
17. In SAP, loads and properties of material are:  
 (A) Fixed data. (B) Output data. (C) Results of the analysis. (D) Not important. (E) Input data.
18. In SAP, self-weight loading always acts in the ... direction.  
 (A) X (B) Y (C) -Y (D) Z (E) -Z
19. Wind load applied over the height of high rise buildings is assumed:  
 (A) Constant. (B) 1000 kN/m. (C) Zero. (D) Parallel to the surface. (E) Perpendicular to the surface.
20. When the material properties are independent of the coordinates, the material is:  
 (A) Isotropic. (B) Non-linear. (C) Plastic. (D) Wood. (E) Homogeneous.

**Question (2): (20 Marks)**

For the shown frame, using the stiffness method:  
 Neglect axial deformation

- (a) Determine the displacements at the nodes due to the given loads.  
 (b) Draw the bending moment diagram.

Note that  $E$  and  $A$  are constants, and the relative moments of inertia are given between brackets.



$$[K_e] = \begin{bmatrix} \left( \frac{EA}{L} \lambda^2 + \frac{12EI}{L^3} \mu^2 \right) & \left( \frac{EA}{L} \mu \lambda - \frac{12EI}{L^3} \mu \lambda \right) & -\frac{6EI}{L^2} \mu & \left( -\frac{EA}{L} \lambda^2 - \frac{12EI}{L^3} \mu^2 \right) & \left( -\frac{EA}{L} \mu \lambda + \frac{12EI}{L^3} \mu \lambda \right) & -\frac{6EI}{L^2} \mu \\ \left( \frac{EA}{L} \mu \lambda - \frac{12EI}{L^3} \mu \lambda \right) & \left( \frac{EA}{L} \mu^2 + \frac{12EI}{L^3} \lambda^2 \right) & \frac{6EI}{L^2} \lambda & \left( -\frac{EA}{L} \mu \lambda + \frac{12EI}{L^3} \mu \lambda \right) & \left( -\frac{EA}{L} \mu^2 - \frac{12EI}{L^3} \lambda^2 \right) & \frac{6EI}{L^2} \lambda \\ -\frac{6EI}{L^2} \mu & \frac{6EI}{L^2} \lambda & \frac{4EI}{L} & \frac{6EI}{L^2} \mu & -\frac{6EI}{L^2} \lambda & \frac{2EI}{L} \\ \left( -\frac{EA}{L} \lambda^2 - \frac{12EI}{L^3} \mu^2 \right) & \left( -\frac{EA}{L} \mu \lambda + \frac{12EI}{L^3} \mu \lambda \right) & \frac{6EI}{L^2} \mu & \left( \frac{EA}{L} \lambda^2 + \frac{12EI}{L^3} \mu^2 \right) & \left( \frac{EA}{L} \mu \lambda - \frac{12EI}{L^3} \mu \lambda \right) & \frac{6EI}{L^2} \mu \\ \left( -\frac{EA}{L} \mu \lambda + \frac{12EI}{L^3} \mu \lambda \right) & \left( -\frac{EA}{L} \mu^2 - \frac{12EI}{L^3} \lambda^2 \right) & -\frac{6EI}{L^2} \lambda & \left( \frac{EA}{L} \mu \lambda - \frac{12EI}{L^3} \mu \lambda \right) & \left( \frac{EA}{L} \mu^2 + \frac{12EI}{L^3} \lambda^2 \right) & -\frac{6EI}{L^2} \lambda \\ -\frac{6EI}{L^2} \mu & \frac{6EI}{L^2} \lambda & \frac{2EI}{L} & \frac{6EI}{L^2} \mu & -\frac{6EI}{L^2} \lambda & \frac{4EI}{L} \end{bmatrix}$$

Where,  $\lambda = \cos \alpha$  and  $\mu = \sin \alpha$

**Question (3): (20 Marks)**

For the shown truss, using the stiffness method:

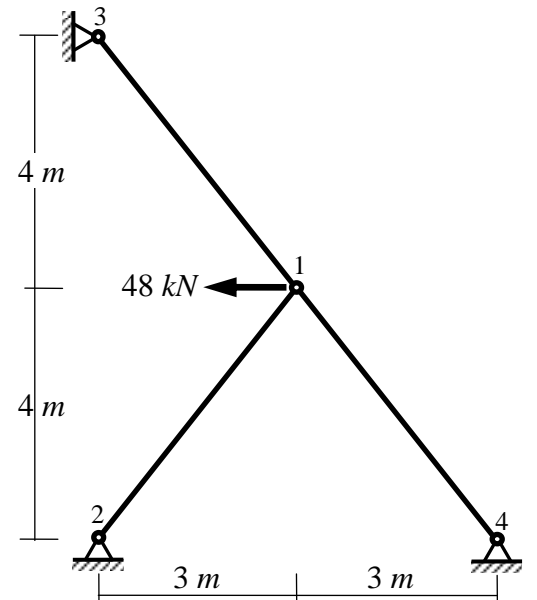
- (a) Determine the displacements at the nodes due to the given loads.  
 (b) Determine the reactions at the supports.  
 (c) Determine the forces in the members.

**Given Data:**

$EA$  is constant for all members.

$$[K_e] = \begin{bmatrix} \frac{EA}{L} \lambda^2 & \frac{EA}{L} \mu \lambda & -\frac{EA}{L} \lambda^2 & -\frac{EA}{L} \mu \lambda \\ \frac{EA}{L} \mu \lambda & \frac{EA}{L} \mu^2 & -\frac{EA}{L} \mu \lambda & -\frac{EA}{L} \mu^2 \\ -\frac{EA}{L} \lambda^2 & -\frac{EA}{L} \mu \lambda & \frac{EA}{L} \lambda^2 & \frac{EA}{L} \mu \lambda \\ -\frac{EA}{L} \mu \lambda & -\frac{EA}{L} \mu^2 & \frac{EA}{L} \mu \lambda & \frac{EA}{L} \mu^2 \end{bmatrix}$$

Where,  $\lambda = \cos \alpha$  and  $\mu = \sin \alpha$



With my best wishes

Dr. M. Abdel-Kader